



**Question Paper For Internal Assessment Examination (Theory) - Credit 4 / 37 /**

**Instructions for Students/Faculty Mid Term I (Total 80 Marks, 2 HRS. Syllabus from Unit-1)**

- Part A: Total number of questions to be given are ten (5 from CO1 and 5 from CO2), each carrying 2 marks and are compulsory to attend. There is no choice. They are short answer type questions (**Not More Than 25 Words For both Question & Answer**), no objective type or fill in the blanks. Total 20 marks.
- Part B: Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student must answer four (2 from CO1 and 2 from CO2). They are long answer type (**Not More Than 50 Words for Question**), each carrying 5 marks. Total 20 marks.
- Part C: Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student must answer four (2 from CO1 and 2 from CO2). They are numerical answer type / fully elaborative type (**Not More Than 70 Words for Question**) \*, each carrying 10 marks. Total 40 marks.

**Mid Term II (Total 120 Marks, 2.5 HRS., Syllabus from Unit-2)**

- Part A: Total number of questions to be given are ten (5 from CO3 and 5 from CO4), each carrying 4 marks and are compulsory to attend. There is no choice. They are short answer type questions (**Not More Than 25 Words For both Question & Answer**), no objective type or fill in the blanks. Total 40 marks.
- Part B: Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student has to answer four (2 from CO3 and 2 from CO4). They are long answer type (**Not More Than 50 Words for Question**), each carrying 7 marks. Total 28 marks.
- Part C: Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student has to answer four (2 from CO3 and 2 from CO4). They are numerical answer type / fully elaborative type (**Not More Than 70 Words For Question**) \*, each carrying 13 marks. Total 52 marks.

**Mid Term III (Total 120 Marks, 2.5 HRS., Syllabus from Unit-3)**

- Part A: Total number of questions to be given are ten (5 from CO5 and 5 from CO6), each carrying 4 marks and are compulsory to attend. There is no choice. They are short answer type questions (**Not More Than 25 Words For both Question & Answer**), no objective type or fill in the blanks. Total 40 marks.
- Part B: Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student must answer four (2 from CO5 and 2 from CO6). They are long answer type (**Not More Than 50 Words for Question**), each carrying 7 marks. Total 28 marks.
- Part C: Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student must answer four (2 from CO5 and 2 from CO6). They are numerical answer type / fully elaborative type (**Not More Than 70 Words for Question**) \*, each carrying 13 marks. Total 52 marks.

\* LIST OF ELABORATIVE THEORY QUESTION SUBJECTS: NO SUBJECT UNDER CREDIT FOUR

**Instructions For Faculties:**

There should be total 6 Course Outcomes (COs) for each subject.

- Mid Term Question Papers are to be submitted as per Course Outcomes (COs) which should be divided equally in Part A, Part B and Part C according to Mid Term Examination and Credit Point.
- In Mid Term-1, the questions are to be given from CO1 and CO2. In Mid Term-2, the questions are to be given from CO3 and CO4. Similarly, in Mid Term-3, the questions are to be given from CO5 and CO6.



- **FACULTY MEMBERS, PLEASE ENSURE EXCEPT ABOVE LISTED SUBJECTS, NO THEORITICAL ELABORATIVE QUESTION SHOULD BE GIVEN IN PART 'C' OF QUESTION PAPER**

**INSTRUCTION FOR STUDENTS**

- **STUDENT IS ALLOWED TO ENTER LATE NOT MORE THAN 15 MIN AFTER STARTING OF EXAM, AND MAY LEAVE THE EXAM HALL ON EXPIRY OF ATLEAST OF 1 Hr FROM THE STARTING TIME OF EXAMINATION.**

**QUESTION PAPER AND STUDENTS DETAILS**

<b>Type of Exam</b>	Mid Term 1	<b>Date of Submission</b>	24/06/2021
<b>Name of Faculty</b>	Mr. Maris Brightson	<b>Date of Examination</b>	29/06/2021
<b>Course</b>	B.Tech (Aeronautical Engineering)	<b>Semester</b>	SEMESTER : 4
<b>Batch</b>	Combined Batches 18, 19, SF 2	<b>Subject</b>	4 AN4 - 04 Aerodynamics-I (Cr 4)

**COURSE OUTCOMES FOR REFERENCE TO FRAME QUESTION PAPER**

(Faculties are required to mention relevant Course Outcome number against the respective question in QP)

<b>Course Outcome</b>	CO1: Distinguish different types of fluid, properties, and potential flow theory in various conditions. CO2: Apply scientific method strategies to analyze the Two-dimensional In-viscid Incompressible flow qualitatively and quantitatively.		
<b>Email I'd</b>	marisbrightson@soaneemrana.org	<b>Phone No.</b>	805-667-7643
<b>Student Name</b>		<b>Student Reg No.</b>	

**Part A**

All the questions are compulsory to attend.

<b>1. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.</b>	1		
<b>Question : 1</b>	Define boundary layer.		
4	Structure of boundary layer	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 2</b>	Define shape parameter.		
5	Momentum Equation for boundary layers	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 3</b>	Define streamline. Write its equation.		



6	Streamlines & Stream function	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 4</b>	Define stream function. Write its equation.		
6	Streamlines & Stream function	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 5</b>	Define velocity potential function. Write its equation.		
7	Potential function	Fundamentals of Aerodynamics - J D Anderson	
<b>2. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.</b>			2
<b>Question : 6</b>	Define vorticity. Write its equation.		
8	Circulation	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 7</b>	Define circulation. Write its SI unit.		
8	Circulation	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 8</b>	Define doublet flow.		
10	Doublet flow	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 9</b>	Define half-oval rankine body.		
11	Half-oval rankine body.	Fundamentals of Aerodynamics - J D Anderson	
<b>Question : 10</b>	Define D'Alembert's paradox.		
12	D'Alembert's paradox	Fundamentals of Aerodynamics - J D Anderson	

### Part B

**FOR MIDTERM 1 - Part B:** Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student must answer four (2 from CO1 and 2 from CO2).

**FOR MIDTERM 2 - Part B:** Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student must answer four (2 from CO3 and 2 from CO4).

**FOR MIDTERM 3 - Part B:** Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student has to answer four (2 from CO5 and 2 from CO6).



**3. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.**

1

**Question : 1**

Derive the continuity equation of the fluid flow in integral and differential form.

2

Continuity Equation (Integral and differential form)

Fundamentals of Aerodynamics - J D Anderson

**Question : 2**

Derive the momentum equation of fluid flow in integral form and deduce it to differential form.

3

Momentum Equation (Integral and differential form)

Fundamentals of Aerodynamics - J D Anderson

**Question : 3**

Explain the following with neat illustrative diagrams  
(1) Boundary layer thickness  
(2) Displacement thickness  
(3) Momentum thickness

4

Structure of boundary layer

Fundamentals of Aerodynamics - J D Anderson

**4. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.**

2

**Question : 4**

Derive the general equation of stream function.

6

Streamlines & Stream function

Fundamentals of Aerodynamics - J D Anderson

**Question : 5**

Derive the relationship between stream function and velocity potential function.

7

Velocity potential function & Equipotential lines

Fundamentals of Aerodynamics - J D Anderson

**Question : 6**

Derive the stream function and velocity potential function for the source flow.

9

Source flow, Sink flow, Combination of source and sink flow

Fundamentals of Aerodynamics - J D Anderson

**Question : 7 (Old Pattern)**

**Part C**



**FOR MIDTERM 1 - Part C:** Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student must answer four (2 from CO1 and 2 from CO2).

**FOR MIDTERM 2 - Part C:** Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student must answer four (2 from CO3 and 2 from CO4).

**FOR MIDTERM 3 - Part C:** Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student has to answer four (2 from CO5 and 2 from CO6).

**5. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.**

1

**Question : 1**

Derive the momentum integral equation for the boundary layer.

5

Momentum Equation for boundary layers

Fundamentals of Aerodynamics - J D Anderson

**Question : 2**

In a 2D incompressible flow the fluid velocity components are given by  $u = x-4y$  and  $v = -y-4x$ . Show that velocity potential function exists and determine its form. Find the equation of stream function.

8

Governing equation for potential flow theory

Fundamentals of Aerodynamics - J D Anderson

**Question : 3**

A source and sink of strength 4 sq.m/s and 8 sq.m/s is located at (-1,0) and (1,0) respectively. Determine the velocity and stream function at a point (1,1) which is lying on the flow field of resultant streamline.

10

Superposition of elementary flows

Fundamentals of Aerodynamics - J D Anderson

**6. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.**

2

**Question : 4**

Imagine that you are standing at a location 6 m from the center of the vortex, and you are feeling 160 km/hr wind. Determine the strength of the vortex.

11

Vortex flow

Fundamentals of Aerodynamics - J D Anderson

**Question : 5**

Derive the expression of pressure coefficient, lift and drag for flow over a non-rotating cylinder.

12

Flow over a non-rotating circular cylinder

Fundamentals of Aerodynamics - J D Anderson

**Question : 6**

Derive the expression of pressure coefficient, lift and drag for flow over a rotating cylinder.

13

Flow over a rotating circular cylinder

Fundamentals of Aerodynamics - J D Anderson



**School of Aeronautics (Neemrana)**

APPROVED BY DIRECTOR GENERAL OF CIVIL AVIATION, MINISTRY OF CIVIL AVIATION, GOVT. OF INDIA  
APPROVED BY ALL INDIA COUNCIL FOR TECHNICAL EDUCATION & AFFILIATED TO RAJASTHAN TECHNICAL UNIVERSITY, KOTA  
& BIKANER TECHNICAL UNIVERSITY, BIKANER, RUN & MANAGED BY L. N. VERMA MEMORIAL SOCIETY

**School of Aeronautics**

APPROVED BY DIRECTOR GENERAL OF CIVIL AVIATION, MINISTRY OF CIVIL AVIATION, GOVT. OF INDIA  
RUN AND MANAGED BY LAXMI NARAIN VERMA MEMORIAL SOCIETY, REGISTERED,  
DELHI ADMINISTRATION, UNDER SOCIETIES REGISTRATION ACT XXI OF 1860.



**Upload Scanned Document In Case of Numerical or Diagram For Any of The Above Questions. (Mention question number with relevant fig / numerical / equations. Max 150 KB)**

**I have scrutinized the question paper. There is no spelling mistake or any type of irrelevant question.**

**Corporate Office : H 974, Palam Extension, Part 1, Sector 7, Dwarka, New Delhi 110077**  
Ph. 011-25084354, 9811315363, 9314009020, E-Mail: [info@soaneemrana.org](mailto:info@soaneemrana.org), [ccashoka@gmail.com](mailto:ccashoka@gmail.com)  
Website: [www.soaneemrana.org](http://www.soaneemrana.org), [www.soaneemrana.org](http://www.soaneemrana.org), [www.soadelhi.com](http://www.soadelhi.com)

The message has been sent from 115.242.250.134 (India) at 2021-06-25 10:20:41 on Firefox 89.0  
Entry ID: 37