### NAME OF STUDY CENTER: SCHOOL OF AERONAUTICS, NEEMRANA

### **Instructions for Students / Faculty**

### Mid Term I (Total 60 Marks, 2 HRS. Syllabus from Unit-1)

- Part A: Total number of questions to be given are six (3 from CO1 and 3 from CO2), each carrying 2 marks and are compulsory to attend. There is no choice. They are short answer type questions (Not More Than 25 Words for Both Question & Answer), no objective type or fill in the blanks. Total 12 marks.
- Part B: Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student has to answer four (2 from CO1 and 2 from CO2). They are long answer type (**Not More Than 50 Words for Question**), each carrying 4 marks. Total 16 marks.
- Part C: Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student has to answer four (2 from CO1 and 2 from CO2). They are numerical answer type / fully elaborative type (**Not More Than 70 Words For Question)**\*, each carrying 8 marks. Total 32 marks.

### Mid Term II (Total 90 Marks, 2.5 HRS., Syllabus from Unit-2)

- Part A: Total number of questions to be given are ten (5 from CO3 and 5 from CO4), each carrying 3 marks and are compulsory to attend. There is no choice. They are short answer type questions (Not More Than 25 Words for Both Question & Answer), no objective type or fill in the blanks. Total 30 marks
- Part B: Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student has to answer four (2 from CO3 and 2 from CO4). They are long answer type (**Not More Than 50 Words for Question**), each carrying 6 marks. Total 24 marks.
- Part C: Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student has to answer any four (2 from CO3 and 2 from CO4). They are numerical answer type / fully elaborative type (**Not More Than 70 Words For Question)**\*, each carrying 9 marks. Total 36 marks.

### Mid Term III (Total 90 Marks, 2.5 HRS., Syllabus from Unit-3)

- Part A: Total number of questions to be given are ten (5 from CO5 and 5 from CO6), each carrying 3 marks and are compulsory to attend. There is no choice. They are short answer type questions (Not More Than 25 Words for Both Question & Answer), no objective type or fill in the blanks. Total 30 marks
- Part B: Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student has to answer four (2 from CO5 and 2 from CO6). They are long answer type (**Not More Than 50 Words for Question**), each carrying 6 marks. Total 24 marks.
- Part C: Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student has to answer four (2 from CO5 and 2 from CO6). They are numerical answer type / fully elaborative type (**Not More Than 70 Words For Question**)\*, each carrying 9 marks. Total 36 marks.
- \* LIST OF ELABORATIVE THEORY QUESTION SUBJECTS: 3 MH4 07 Manufacturing Process, 4 AN4 06 Aircraft Materials and Processes (Cr 3), 5 AN4 05 Aircraft System (Cr 3), 6 AN4 05 Avionics-I (Cr 3), 6 MH4 03 Applied Hydraulics & Pneumatics (Cr 3), 6 MH5 11 Principles of Management (Cr 3), 6 MH5 13 Aircraft Electronics System (Cr 3), 7 AN5 12 Maintenance of Airframe and System (Cr 3), 7 AN5 13 Helicopter Theory (Cr 3), 7 AG6 60.1 Human Engineering and Safety (Cr 3), 7 ST 01 Avionics II (Special Theory Subject) (Cr 3), 7 MH5 11 Design of Mechatronics Systems (Cr 3), 7 MH5 12 Robotics and Machine Vision System (Cr 3), 7 MH6 13 Medical Electronics (Cr 3), 7 AN6 60.1 Aircraft Avionic System (Cr 3), 8 AN5 12 Maintenance of Power Plant and System





# NAME OF STUDY CENTER: SCHOOL OF AERONAUTICS, NEEMRANA

(Cr 3), 8 AN5 - 13 Unmanned Aerial Vehicles & Systems (UAV) (Cr 3), 8 MH5 - 13 Product Development & Launching (Cr 3), 8 EC6 - 60.2 Robotics and control (Cr 3)

#### **Instructions For Faculties**

There should be total 6 Course Outcomes (COs) for each subject.

- Mid Term Question Papers are to be submitted as per Course Outcomes (COs) which should be divided equally in Part A, Part B and Part C according to Mid Term Examination and Credit Point.
- In Mid Term-1, the questions are to be given from CO1 and CO2. In Mid Term-2, the questions are to be given from CO3 and CO4. Similarly, in Mid Term-3, the questions are to be given from CO5 and CO6.
- FACULTY MEMBERS, PLEASE ENSURE EXCEPT ABOVE LISTED SUBJECTS, NO THEORITICAL ELABORATIVE QUESTION SHOULD BE GIVEN IN PART 'C' OF QUESTION PAPER

#### INSTRUCTION FOR STUDENTS

STUDENT IS ALLOWED TO ENTER LATE NOT MORE THAN 15 MIN AFTER STARTING OF EXAM,

QUESTION PAPER & STUDENTS DETAILS			
Type of Exam	Internal Improvement Exam Date of Submission 22/03/2021		22/03/2021
Name of Faculty	Mr. Korapati Akhil	Date of Examination	26/03/2021
Course	B.Tech (Aeronautical Engineering)	Semester	SEMESTER: 5
Batch	Combined Batches 12, 13, 14	Subject	5 AN4 - 04 Propulsion-I (Cr 3)-
COURSE OUTCOMES FOR REFERENCE TO FRAME QUESTION PAPERS (Faculties are required to mention Course Outcome Number against each part of the question paper)			
Course Outcome	CO 4. Investigate the Propeller Pitch Control and Propeller Synchronizing in aircraft power plant.  CO 5. Demonstrate about the Propeller Maintenance procedure.  CO 6. Explain the working of Gas Turbine Engines basic principle and performance parameters.  Mail I'd k.akhil2110@gmail.com  Phone No.  701-345-8080		
Email I'd			
Student Name			
PART A			

All the questions are compulsory to attend.





# NAME OF STUDY CENTER: SCHOOL OF AERONAUTICS, NEEMRANA

TO A STORY OF THE			
	CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.		
Question : 1	Classify the types of propulsion system based on engines?		
1	Objective of propulsion	IC engine V Ganesan	
Question : 2	Explain about the Heat engine and	classify based on engine r	mechanism?
2	Definition of engine and working process	IC engine V Ganesan	
Question : 3	Draw ideal P-V and T-S diagrams for 4 - stoke Otto cycle and also 4- stroke diesel cycle engines.		
11	Compression ratio and engine efficiencies	IC engine V Ganesan	
Question : 4	Discuss about difference between	SI & CI engines.	
5	SI & CI engines.	IC engine V Ganesan	
Question : 5	stion: 5 Explain about specific fuel consumption of reciprocating engines?		
9	Engine performance and parameters	IC engine V Ganesan	
	DUTCOME (CO) NUMBER ACCO	PRDING TO THE TYPE	
Question : 6	Classify the Basic thrust providing	propellers.	
15	Fundamentals of propellers	Fundamentals of propellers kroes	
Question : 7	How the propeller torque reaction a	acting on the aircraft?	
16	thrust force and torque	Fundamentals of propellers kroes	
Question : 8	Explain about impact damage, ove	r speed damage?	
28	impact damage, over speed damage	Fundamentals of propellers kroes	
Question : 9	Discuss about slip and slip function	٦.	
15	propeller slip	Fundamentals of propellers kroes	
Question : 10	How to find stages of compressor i	n jet engine ?	
36	construction arrangement of jet engines	Fundamentals of propellers kroes	
PART B		'	





# NAME OF STUDY CENTER: SCHOOL OF AERONAUTICS, NEEMRANA

**FOR MIDTERM 1 - Part B**: Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student must answer four (2 from CO1 and 2 from CO2).

**FOR MIDTERM 2 - Part B**: Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student must answer four (2 from CO3 and 2 from CO4).

**FOR MIDTERM 3 - Part B**: Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student has to answer four (2 from CO5 and 2 from CO6).

3. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.			
Question : 1	Discuss the difference between id	Discuss the difference between ideal and actual valve timing diagrams of petrol engine ?	
8	valve timing diagram	IC engines RK RAJPUT	
Question : 2	Explain engine terminologies:- 1) Bore. 2) stroke. 3) stroke to Bore ratio. 4) dead centers. 5) displacement of swept volume. 6) cubic capacity. 7) clearance volume. 8) compression ratio		
2	working of IC engines	IC engines RK RAJPUT	
Question : 3	Draw and discuss about SI engine	Draw and discuss about SI engine battery ignition system and magneto ignition system	
14	ignition system	IC engines RK RAJPUT	
	RSE OUTCOME (CO) NUMBER ACC PER INSTRUCTIONS ABOVE.	ORDING TO THE TYPE	
Question : 4	Explain about:- 1) Propeller tracking. 2) Static propeller balancing. 3) Dynamic propeller balancing. 4) Engine vibration analysis.		
17	engine propeller vibrations	Fundamentals of propellers kroes	
Question : 5	Draw and Explain operation merit 1) turbo prop engine. 2) turbo jet engine. 3) turbo fan engine. 4) turbo shaft engine	s and demerits of jet engine	s those are :-
36	operation merits and demerits of jet engines	Gas Turbies V ganesan	





NAME OF STUDY CENTER: SCHOOL OF AERONAUTICS, NEEMRANA			
overall efficiency	Gas Turbies V ganesan		
FOR MIDTERM 1 - Part C: Total number of questions to be given are six (3 from CO1 and 3 from CO2), out of which student must answer four (2 from CO1 and 2 from CO2).  FOR MIDTERM 2 - Part C: Total number of questions to be given are six (3 from CO3 and 3 from CO4), out of which student must answer four (2 from CO3 and 2 from CO4).  FOR MIDTERM 3 - Part C: Total number of questions to be given are six (3 from CO5 and 3 from CO6), out of which student has to answer four (2 from CO5 and 2 from CO6).			
5. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.			
In a constant volume 'Otto cycle', the pressure at the end of compression is 15 times that at the start, the temperature of air at the beginning of compression is 38 degrees centigrade and maximum temperature attained in the cycle is 1950 degrees centigrade determine:  1) compression ratio.  2) thermal efficiency of the cycle.  3) work done.			
compression ratio engine efficiencies	IC engines RK RAJPUT		
An engine with 200 mm cylinder diameter and 300 mm stroke works on theoretical diesel cycle . the initial pressure and temperature of air used are 1 bar and 27 degrees. the cut-off is 8 percentage of the stroke determine:  1) pressure and temperatures at all salient points. 2) theoretical air standard efficiency. 3) mean effective pressure.			
compression ratio engine efficiencies	IC engines RK RAJPUT		
Derive Thrust equation and find speed ratio of jet engine?			
thrust equation	GAS turbines V Ganesan		
6. CHOOSE COURSE OUTCOME (CO) NUMBER ACCORDING TO THE TYPE OF MIDTERM, AS PER INSTRUCTIONS ABOVE.			
A turbo jet power plant user aviation kerosene have calorific value of 43 m j /kg fuel consumption is 0.18 kg / hour / N of thrust , when the thrust is 9 KN. the aircraft velocity is 500 m/s the mass of air passing through the compressor is 27 kg/s . calculate the air fuel ratio and overall efficiency			
performance parameters of jet engine	GAS turbines V Ganesan		
	overall efficiency  It C: Total number of questions to be four (2 from CO1 and 2 from CO2 at C: Total number of questions to be four (2 from CO3 and 2 from CO4 at C: Total number of questions to be four (2 from CO5 and 2 from CO4 at C: Total number of questions to be four (2 from CO5 and 2 from CO5 and 2 from CO6 at C: Total number of questions to be four (2 from CO5 and 2 from CO6 at C: Total number of questions to be four (2 from CO5 and 2 from CO6 at C: Total number of questions to be four four (2 from CO5 and 2 from CO6 at C: Total number of questions to be four four four four four four four four	C: Total number of questions to be given are six (3 from or four (2 from CO1 and 2 from CO2).  C: Total number of questions to be given are six (3 from or four (2 from CO3 and 2 from CO4).  C: Total number of questions to be given are six (3 from or four (2 from CO5 and 2 from CO4).  C: Total number of questions to be given are six (3 from or four (2 from CO5 and 2 from CO6).  DUTCOME (CO) NUMBER ACCORDING TO THE TYPE NSTRUCTIONS ABOVE.  In a constant volume 'Otto cycle', the pressure a times that at the start, the temperature of air at the beginning centigrade and maximum temperature attained in the cycle determine:  1) compression ratio.  2) thermal efficiency of the cycle.  3) work done.  compression ratio engine IC engines RK RAJPUT  An engine with 200 mm cylinder diameter and 300 mm strocycle. the initial pressure and temperature of air used are 1 is 8 percentage of the stroke determine:  1) pressure and temperatures at all salient points.  2) theoretical air standard efficiency.  3) mean effective pressure.  compression ratio engine IC engines RK RAJPUT  Derive Thrust equation and find speed ratio of jet engine?  thrust equation  GAS turbines V Ganesan  DUTCOME (CO) NUMBER ACCORDING TO THE TYPE NSTRUCTIONS ABOVE.  A turbo jet power plant user aviation kerosene have calc consumption is 0.18 kg / hour / N of thrust , when the thrust 500 m/s the mass of air passing through the compressor is ratio and overall efficiency  performance parameters of jet GAS turbines V	





Question Paper For Internal Assessment Examination (Theory) - Credit 3 / 97 / SET 1

NAME OF STUDY CENTER: SCHOOL OF AERONAUTICS, NEEMRANA		
Question : 5	The effective jet exist velocity from a jet engine is 2700 m/s the forward flight velocity 1350 m/s and the air flow rate is 78.6 kg / s calculate the:- 1) trust. 2) thrust power. 3) propulsive efficiency.	
40	performance parameters of jet GAS turbines V	

Ganesan

**RAJPUT** 

	A 42.5 KW engine has a mechanical efficiency of 85 percent. Find the indicated power and frictional power. If the frictional power is assumed to be constant with load, what will be the mechanical efficiency at 60 percent?		
0	Engine performance and IC engines RK		

Upload Scanned Document In Case of Numerical or Diagram For Any of The Above Questions. (Mention	
question number with relevant fig / numerical / equations. Max 150 KB)	

parameters

engine

40

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I have scrutinized the question paper. There is no spelling mistake or any type of irrelevant question.