School of Aeronautics (Neemrana) APPROVED BY DIRECTOR GENERAL OF CIVIL AVIATION, MINISTRY OF CIVIL AVIATION, GOVT OF INDIA APPROVED BY ALL NOIA COUNCIL ROR TECHNICA. EDUCATION & AFFILIATE OT BRAINSTHIM TECHNICAL UNIVERSITY, KOTA B BRANER TECHNICAL UNIVERSITY, BIKANER, RUN & MANAGED BY L IN VERMA MEMORIAL SOCIETY

APPROVED BY DRECTOR GENERAL OF CIMILAWATION, MINISTRY OF CIVILAWATION, GOVT. OF INDIA RUN AND MANAGED BY LAXIM NARAIN VERMA MEMORIAL SOCIETY, REGISTERED, DELHI ADMINISTRATION, UNDER SOCIETIES REGISTRATION ACT XXI OF 1860.



Question Paper For Internal Assessment Examination (Theory) - Credit 4 / 13 / SET 1

Instructions For Students / Faculty

Mid Term I (Total 80 Marks, 2 HRS. Syllabus From Beginning Of Session)

- Part A: Total number of questions to be given are ten, each carrying 2 marks and are compulsory to attend. There is no choice. They are short answer type questions (**Not More Than 25 Words For both Question & Answer**), no objective type or fill in the blanks. Total 20 marks.
- Part B: Total number of questions to be given are six, out of which student has to answer any four. They are long answer type (**Not More Than 50 Words For Question**), each carrying 6 marks. Total 24 marks.
- Part C: Total number of questions to be given are four, out of which student has to answer any three. They are numerical answer type / fully elaborative type (**Not More Than 70 Words For Question**)*, each carrying 12 marks. Total 36 marks.

Mid Term II & III (Total 120 Marks, 2.5 HRS. Syllabus From Beginning Of Session)

- Part A: Total number of questions to be given are ten, each carrying 2 marks and are compulsory to attend. There is no choice. They are short answer type questions (**Not More Than 25 Words For both Question & Answer**), no objective type or fill in the blanks. Total 20 marks.
- Part B: Total number of questions to be given are seven, out of which student has to answer any five. They are long answer type (Not More Than 50 Words For Question), each carrying 8 marks. Total 40 marks.
- Part C: Total number of questions to be given are five, out of which student has to answer any four. They are numerical answer type / fully elaborative type (**Not More Than 70 Words For Question**)*, each carrying 15 marks. Total 60 marks.

* LIST OF ELABORATIVE THEORY QUESTION SUBJECTS: NO SUBJECT UNDER CREDIT FOUR

FACULTY MEMBERS, PLEASE ENSURE EXCEPT ABOVE LISTED SUBJECTS, NO THEORITICALELABORATIVE QUESTION SHOULD BE GIVEN IN PART 'C' OF QUESTION PAPER

STUDENT IS ALLOWED TO ENTER LATE NOT MORE THAN 15 MIN AFTER STARTING OF EXAM, AND MAY LEAVE THE EXAM HALL ON EXPIRY OF ATLEAST OF 1 Hr FROM THE STARTING TIME OF EXAMINATION.

Question Paper & Student Details Type of Exam Internal Improvement Exam Date of Submission 25/11/2020 Name of Faculty Dr. Bipin Kumar Dwivedi Date of Examination 03/12/2020 Course B.Tech (Aeronautical Engineering) Semester SEMESTER: 4 04 4 AN4 Batch Subject Aerodynamics-I (Cr 4) COURSE OUTCOMES FOR REFERENCE TO FRAME QUESTION PAPER (Faculties are required to mention relevant Course Outcome number against the respective question in QP) **Course Outcome** 3. School of Aeronautics (Neemrana) Corporate Office : H 974, Palam Extension, Part 1, Sector 7, Dwarka, New Delhi 110077

Ph. 011-25084354, 9811315363, 9314009020, E-Mail: info@soaneemrana.org, ccashoka@gmail.com Website: www.soaneemrana.org, www.soaneemrana.org, www.soadelhi.com

Approved by Dir Approved by Dir a Bikaner techn	L of Aeronautics (Neemrana)	chool of Aeron D by Drector General of CML AWATON, MINISTRY OF CML D MANAGED BY LAXM NARAMIN VERMA MEDINAL SC ADMINISTRATION, UNDER SOCIETIES REGISTRATIO	AMATION GOVT. OF INDIA AVAITION GOVT. OF INDIA N ACT XXI OF 1880.			
Email I'd	bipinkumardwivedi@soaneemrana.org	Phone No.	931-400-9035			
Student Name		Student Reg No.				
Part A						
Question : 1	Differentiate between laminar and turbulen	t boundary layer.				
2	Boundary layer	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON				
Question : 2	What is stream function equation of uniform	m flow?				
4	Potential flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON				
Question : 3	What is doublet flow?					
5	Potential flow FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 4	What is Kutta condition?					
10	Invisid incompressible flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON				
Question : 5	Define Kutta-Joukowski transformation.					
12	Invisid incompressible flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON				
Question : 6	Define thin airfoil theory.	•	•			
16	Thin airfoil theory FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 7	Differentiate between finite and infinite win	g.	*			
20	Finite wing theory FUNDAMEN AERODYNAI JHON D ANE					
Question : 8	What is critical Mach number?	*	-			
24	Finite wing theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON				
Question : 9	Differentiate between aerodynamic center	and centre of pressure.				
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26	Finite wing theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 10	What do you mean by low speed wi	What do you mean by low speed wind tunnel?					
30	wind tunnel	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Part B							
Question : 1	Derive the fundamental equation of	thin airfoil theory.					
12	Thin airfoil theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 2	Prove that the moment coefficient a inviscid, incompressible flow over a of attack 'a' is equal to zero.	bout the quarter-chord point for the thin symmetrical airfoil at small angle					
18	Thin airfoil theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 3	Derive the expression of lift coefficie over a thin cambered airfoil at smal	ent for the inviscid, incompressible flow I angle of attack 'α'.					
15	Thin airfoil theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 4	Explain about the Lifting line theory						
16	Finite wing theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 5	Explain about the Biot and savart la	W.					
20	Finite wing theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 6	Explain the effect of ideal flow ovr ro	Explain the effect of ideal flow ovr rotating cylinder.					
24	Potential flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					
Question : 7	How to calibrate low speed wind tur	nnel? Explain.					
34	wind tunnel	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON					

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Part C

Question : 1	Obtain the equation to the streamlines for the velocity flow field given as, $V= 2x^3i - 6x^2yj$.				
23	Potential flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON			
Question : 2	Consider an NACA 2412 airfoil with a 2-m chord in an airstream with a velocity of 50 m/s at standard sea level conditions. If the lift per unit span is 1353 N, what is the angle of attack?				
26	Finite wing theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON			
Question : 3	Consider a thin, symmetric airfoil at 1.5 angle of attack. From the results of thin airfoil theory, calculate the lift coefficient and the moment coefficient about the leading edge.				
29	Thin airfoil theory	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON			
Question : 4	For a doublet of strength 20m^2/s. Calculate the velocity at point P(1,2) and the value of stream function passing through it.				
23	Invisid incompressible flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON			
Question : 5	A long circular cylinder lies in an air stream having a velocity of 60m/s. There is a flow around the cylinder with circulation -400m^2/s (clockwise). Calculate the lift force per unit of cylinder.				
24	Invisid incompressible flow	FUNDAMENTAL OF AERODYNAMICS BY JHON D ANDESON			
Upload Scanned Docur For Any of The Above O relevant fig / numerical / e	nent In Case of Numerical or Diagram Questions. (Mention question number with equations. Max 150 KB)				
I have scrutinized the mistake or any type of i	question paper. There is no spelling rrelevant question.	Yes			
Corporate Office : H 974, Palam Extension, Part 1, Sector 7, Dwarka, New Delhi 110077 Ph. 011-25084354, 9811315363, 9314009020, E-Mail: info@soaneemrana.org, ccashoka@gmail.com Website: www.soaneemrana.org, www.soaneemrana.org, www.soadelhi.com					

School of Aeronautics (Neemrana)

Question Paper For Back / Re-back Internal Assessment Examination (Theory) - Old Scheme i.e 2012 Syllabus

Instructions For Students / Faculty

Back / Re-back Internal Examination (Total 60 Marks, 2 Hrs, Syllabus From Beginning of The Session)

Total number of questions to be given are 10, each carrying 10 marks and it is compulsory to attend 2 questions from Part A and 4 questions from Part B. There is a choice of two questions out of four in part A and 4 questions out of 6 in Part B. Part A will be theoretical or derivation type (Not More Than 70 Words For Question). Part B will be fully numerically oriented questions (Not More Than 70 Words For Question), except for the list of subjects given below. No objective type or fill in the blanks shall be given, but subpart of question can be given for both Part A & B.

* LIST OF ELABORATIVE THEORY QUESTION SUBJECTS: Aircraft Materials, Aircraft System, Aircraft Rules & Regulation-I, Mechanics of Composite Materials, Aircraft Design, Aircraft Rules & Regulation-II, Avionics-I, Helicopter Theory, Maintenance of Airframe and System Design, Avionics-II, Airlines and Airport Management, Maintenance of Power Plant & Systems

FACULTY MEMBERS, PLEASE ENSURE EXCEPT ABOVE LISTED SUBJECTS, NO THEORITICAL ELABORATIVE QUESTION SHOULD BE GIVEN IN PART 'B' OF QUESTION PAPER

STUDENT IS ALLOWED TO ENTER LATE NOT MORE THAN 15 MIN AFTER STARTING OF EXAM, AND MAY LEAVE THE EXAM HALL ON EXPIRY OF ATLEAST OF 1 Hr FROM THE STARTING TIME OF EXAMINATION

Name of Faculty*		Ashok Bhatia		Date of Submission of QP		26/11/2020		
Subject*	AAN6 - Machine Design (Old)		Date of Examination*		02/12/2020			
Email Id of Faculty:* ashok		bhatia@soaneemrana.org		Course* B.Tech (Aero		nautical Engineering)	•	
Phone Number of Faculty*		798 815 8760		Semester*	* Semester : 4		•	
Student Name					Student Reg l	No.		

Question Paper & Student Details

Part A

Question : 1*	Find the dia be taken as inside and c	Find the diameter of a solid steel shaft to transmit 20 kW at 200 r.p.m. The ultimate shear stress for the steel may be taken as 360 MPa and a factor of safety as 8. If a hollow shaft is to be used in place of the solid shaft, find the inside and outside diameter when the ratio of inside to outside diameters is 0.5.						
					li			
esson Plan*	10	Topic*	Shaft	Source*	Machine Design by R.s.			

Question : 2*	Derive an expressiof four bolts.	ion for the maxii	mum load in a bolt when a bracke	t with circular base is b	olted to a wall by means
Lesson Plan*	6	Topic*	Screwed Joint	Source*	Machine Design by R.s.
Question : 3*	Design a gib and α material and have σt = 20 MPa ; τ = 1	cottor joint to ca the following al 5 MPa ; and σc =	rry a maximum load of 35 kN. Assu lowable stresses : = 50 MPa	uming that the gib, cotte	er and rod are of same
Lesson Plan*	7	Topic*	Cotter Joint	Source*	Machine Design by R.s.
Question : 4*	Write Soderberg's	equation and st	ate its application to different typ	e of loading.	
Lesson Plan*	14	Topic*	Variable Loading	Source*	Machine Design by R.s.
Part B					
Question : 1*	A 150 mm diamete length is 1.5 times viscosity of the oil	er shaft support the shaft diame at the operating	ing a load of 10 kN has a speed of eter. If the diametral clearance of temperature is 0.011 kg/m-s, find	1500 r.p.m. The shaft ru the bearing is 0.15 mm I the power wasted in fr	uns in a bearing whose and the absolute iction.
Lesson Plan*	24	Topic*	Bearings	Source*	Machine Design by R.s.
Question : 2*	Explain the metho the axis of the bol	d of determinin; t.	g the size of the bolt when the bra	acket carries an eccentri	c load perpendicular to

Lesson Plan*	19	Topic*	Screwed Joint	Source*	Machine Design by R.s.			
Question : 3*	A steel solid shaft transmitting 15 kW at 200 r.p.m. is supported on two bearings 750 mm apart and has two gears keyed to it. The pinion having 30 teeth of 5 mm module is located 100 mm to the left of the right hand bearing and delivers power horizontally to the right. The gear having 100 teeth of 5 mm module is located 150 mm to the right of the left hand bearing and r							
Lesson Plan*	11	Topic <mark>*</mark>	Shaft	Source*	Machine Design by R.s.			
Question : 4* A simply supported beam has a concentrated load at the centre which fluctuates from a value of P to 4 P. The span of the beam is 500 mm and its cross-section is circular with a diameter of 60 mm. Taking for the beam material an ultimate stress of 700 MPa, a yield stress of 500 MPa, endurance limit of 330 MPa for reversed bending, and a factor of safety of 1.3, calculate the maximum value of P. Take a size factor of 0.85 and a surface finish factor of 0.9.								
Lesson Plan*	7	Topic*	Variable Loading	Source*	Machine Design by R.s.			
Question : 5	Explain the gen	eral consideratio	ns in machine design.		1			
Lesson Plan	2	Торіс	Design Fundamentals	Source	Machine Design by R.s.			
Question : 6 Select a single row deep groove ball bearing for a radial load of 4000 N and an axial load of 5000 N, operating at a speed of 1600 r.p.m. for an average life of 5 years at 10 hours per day. Assume uniform and steady load.								
Lesson Plan	26	Торіс	Bearings	Source	Machine Design by R.s.			
Upload Scanned Doc Case of Numerical or for any of the above Mention question number relevant fig / numerical /	Iload Scanned Document In se of Numerical or Diagram r any of the above question Intion question number with evant fig / numerical / equations.							

Max 150 KB

I have scrutinized the question paper. There is no spelling mistake or any type of irrelevant question.